



P92 ECHO MKII 100HP Standard Conf gurat on G3X Avionic suite Base Price

GSU CONFIGURATION	
AMPS (AMMETER SHUNT OR HALL EFFECT)	MANIFOLD PRESSURE
MONITOR CHT	FUEL PRESSURE
AIRCRAFT BUS VOLTAGES	RPM
OIL TEMPERATURE	TRIM INDICATOR
OAT	LH + RH FUEL QTY
OIL PRESSURE	
AVIONIC PACKAGE for G3X configuration:	
GARMIN GMA 245 Audio Panel	Antennas:
GARMIN GTR 225A COM	Transponder
GARMIN GTX 35R REMOTE TRANSPONDER with GPS	VHF
KANNAD ELT 406 Mhz WITH REMOTE MOUNTED SWITCH	ELT
STICK PUSH-TO-TALK SWITCH-PILOT/COPILOT	
MIC & PHONE JACKS-PILOT/COPILOT	
US LSA	
Extended Warranty for Rotax Engine (1 year extension)	Baggage Compartment Net
Fire Wall Inox	External Power Supply
Tie Down Rings	Taxi Light LED
Fire Sleeve On Fuel and Oil Tubes	Wing Light Green/Red, White and Strobe AVEO
Thermostatic Valve Oil	Windshield Defroster

Packing \$ 1,891.00

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GTX 45R TDX mode S ADS-B IN/OUT Remote mounted,	
INCLUDES GPS20A & GA37 (Only with G3X) (Exch. For Std. GTX35R)	\$ 4,300.00
Autopilot 2 Axis for G3X	\$ 4,550.00
GMC 307 Additional Autopilot Mode Controller	\$ 2,670.00
P92 Special Paint Two Colors	\$ 4,750.00

OTHERS & ACCESSORIES

Specifications, weights, representations, colors, list of equipment, use of materials and model references listed herein are not warranted or guaranteed to be true or accurate

Standard equipment

Two Display G3X Touch with EIS

- GDU460
- GDU465
- ADAHRS GSU 25
- GEA 24 Engine Instrument Module
- GMU 22 Magnetometer
- GTP 59 Temperature Probe
- LRU KIT
- INSTALLATION KIT
- G3X SENSOR KIT
- GA 56 Antenna
- GA37 Antenna (ADSB & XM)
- GPS A20 - ADSB

Flight Instruments and Indicators

- Magnetic Compass
- Airspeed Ind
- Altimeter
- Vertical Speed
- Bank Indicator
- Pitot System
- Static System
- Stabilator Trim Position Indicator

Engine Instruments & Indicators

- Tachometer
- Hour Recorder
- Oil Press
- Oil Temp
- Head Temp
- Fuel Press
- Voltmeter
- Lh + Rh Fuel Qty
- Stabilator Trim Position Indicator
- Flap Indicator

Flight controls

- Hydraulic Toe Brakes
- Parking Brake
- Electrical Flaps
- Dual Flight Controls
- Steerable Nose Wheel
- Stabilator Trim (Electric Actuated From Stick)
- Engine Controls:
 - Central Quadrant with Single Throttle Lever
 - Choke
- Flight Trim Controls:
 - Stabilator With Indicator
- Fuel Control Selector Andair

Panel switches

- Starter Key ACS
- Split Starter

Electrical system

- 12 Volt 18A Amp. Battery
- 12 Volt Alternators-20 Amp
- Circuit Breaker Panel
- 12 Volt Socket

Fuel system

- Two Integral Fuel Tanks With 95Lt/25Usgal Total Capacity
- Aux Fuel Pump, electric
- Engine Driven Fuel Pump
- Fuel Tank Quick Drain, Two

Interiors

- Pilot And Copilot Seats:
 - Adjustable Fore And Aft
- Seat Belts
- Wall To Wall Carpeting
- Map and Storage Pockets
- Luggage Compartments

Exterior

- Epoxy Corrosion Proofing, All Structure
- Lh/Rh Front Door Pilot/Copilot, Lock And Key
- All Lateral And Rear Windows Tinted Brown
- Aluminum MLG, wheel track increased
- Main Wheels, 5,00 X 5
- Nose Wheel, 5,00 X 5

Exteriors lights

- Landing Light (No Led)

Cabin comfort system

- Heating System
- Ventilator Adjustable 2 Place

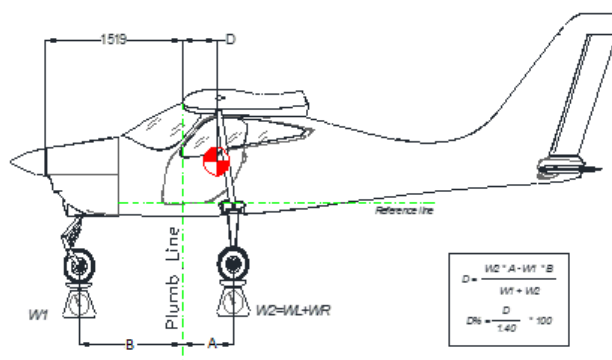
Powerplant and Propeller

- Engines Rotax 912ULS2 100 Hp, 4 Cylinders
- Liquid/Air Cooled, Integrated Reduction Gear
- Dual Ignition System
- Throttle Control
- Tubular Steel Engine Mount
- Propeller Sensenich, 2 Blades, Fixed
- Propeller Spinner
- Air Filter
- Oil Filter
- Oil And Water Coolers

Product Support and Documents

- Manufacturer's Full Two Year Limited Warranty
- Pilots Operation Handbook
- Maintenance Manual
- Parts Catalog

 TECNAM COSTRUZIONI AERONAUTICHE	Aircraft: P92MK2	S/N: 1673	Mark: N5673T	Date: 27/01/22	Report N°: P92MK2-1673
Manufactured by: Costruzioni Aeronautiche TECNAM S.p.A. – Via Majorise (CE) – Italy Tel.: 0823.62.01.34					



Datum : leading edge vertical

Equipment list : Date 27/01/22

WL = Balance value MLG (LH):.....			344,0	Lb
WR = Balance value MLG (RH):.....			342,0	Lb
WN = Balance value NLG (WN):.....			133,0	Lb
WP=WL+WR:.....			686,0	Lb
WTOT=WP+WN:.....			819,0	Lb
USEFUL LOAD=MTOW-WTOT:.....			501,0	Lb
Distance <A>:.....			22,2	Inch
Distance :.....			44,9	Inch
Mean Aerodynamic chord (CMA):.....			55	Inch
Distance C.G. from DATUM:.....			D=[WP*(A)-[WN*(B)] Wtot	
Momentum:	58398	Lb*in	D:.....	11,3 Inch
C.G. % From chord:.....			D%= $\frac{D}{1400}$	* 100
D:.....			20,6	%

Remark: The empty weight enclosed unusable fuel (1.55 Liter for each fuel tanks) .

The Empty weight enclosed oil engine , coolant liquid, oil brake.

Control: _____